

Vergleichende Bewertung der Paketdistribution mit Drohnen und Lieferwagen

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82. Jahrestagung des VHB

Frankfurt, 19.03.2020



Outline

- 1 Introduction
- 2 Energy consumption of DVs and EVs
- 3 Energy consumption of UAVs
- 4 Simulation experiments
- 5 Conclusion

Motivation



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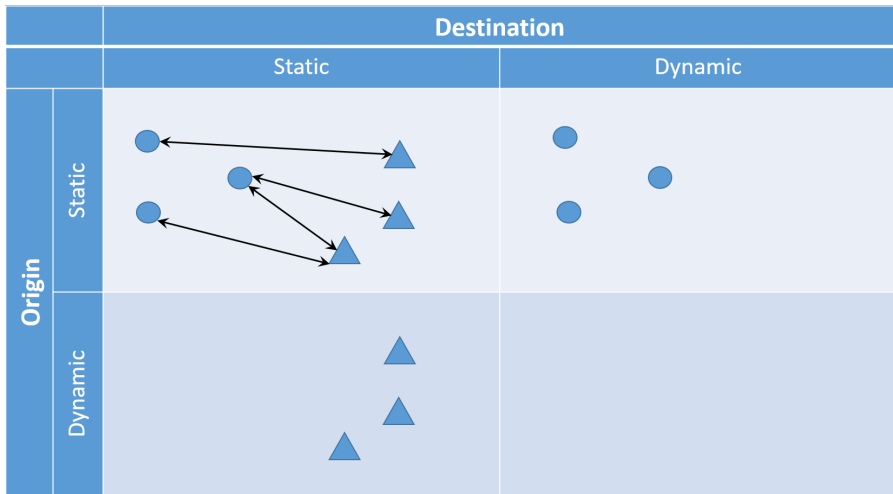
Perceived advantages:

- fast & flexible → same-day deliver within 30 minutes [Amazon, 2019]
- cost efficient → 1 \$ per delivery [Wang, 2016], 1-5 ct per mile [Peers, 2018]
- green → no/less GHG emissions than trucks [Goodchild and Toy, 2018]
- save → less accidents, less congestion [Crowe, 2019]

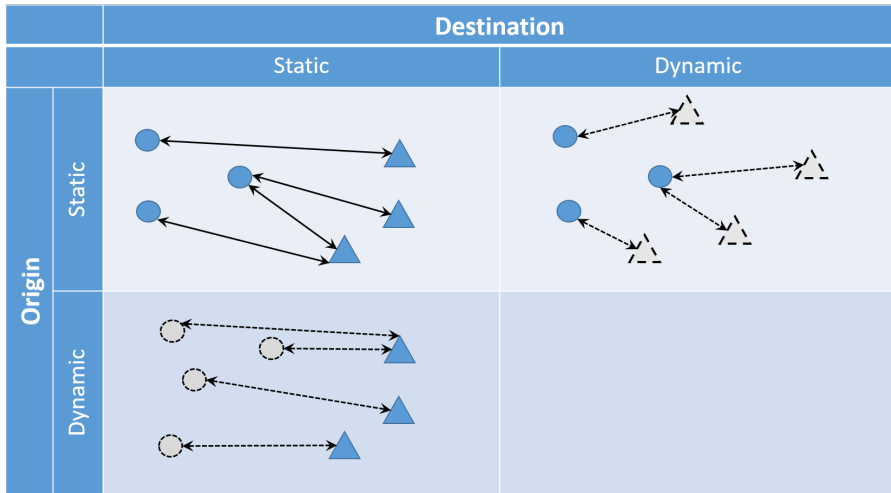
Current status:

- some pilot projects [e.g. blood samples, Scott et al., 2017]
- Alphabet Wing drones received regulatory permission in US and Australia [Lee, 2019]
- Amazon Prime Air expected to receive permission in 2019 [Lee, 2019]

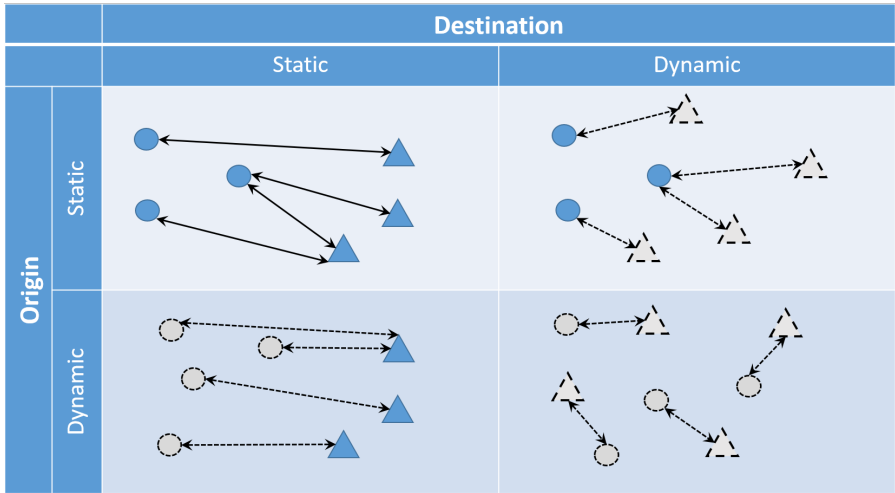
Application scenarios



Application scenarios



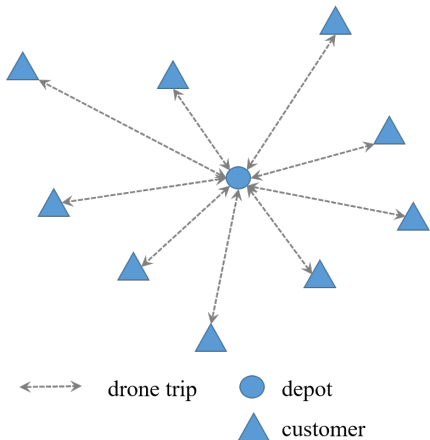
Application scenarios



Application scenarios

		Destination	
		Static	Dynamic
Origin	Static		
	Dynamic		

Performance of stationary drone delivery systems



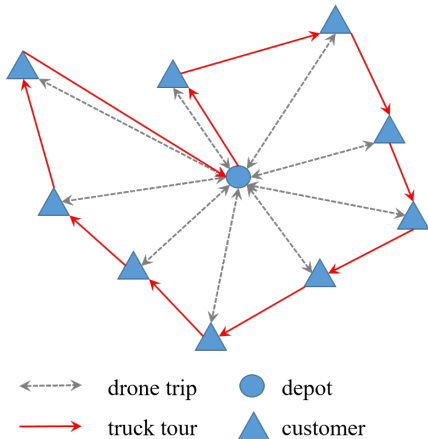
Assumptions:

- fixed depot & customer locations
- technological specifications
- drone capacity: 1 parcel

KPIs:

- service time
- investment cost
- operating cost
 - operator
 - wear-&-tear
 - energy
- emissions

Performance of stationary drone delivery systems



⇒ comparison of energy demand and associated emissions between trucks and drones

Assumptions:

- fixed depot & customer locations
- technological specifications
- drone capacity: 1 parcel

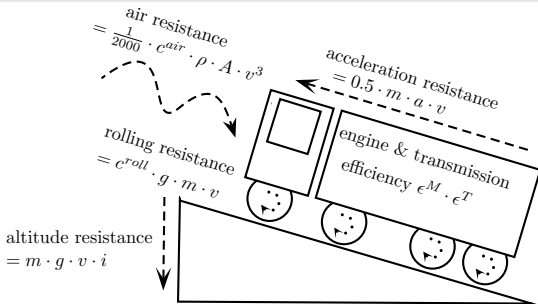
KPIs:

- service time
- investment cost
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 - operator
 - wear-&-tear
 - **energy**
- **emissions**

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Power demand & energy consumption

Demir et al. [2014], Kirschstein and Meisel [2015]



$$P^{EDV} = P^{roll} + P^{air} + P^{climb} + P^{acc} + P^{aux}$$

$$E^{DV} = \left(t \cdot \left(f^{idle} + \frac{f^{full} - f^{idle}}{\epsilon_{DV}^T(v) \cdot P} \cdot P^{EDV} \right) \cdot N_{Diesel} \right) \cdot \frac{1}{\epsilon_{Diesel}^{wtt}}$$

$$E^{EV} = t \cdot \frac{P^{EDV}}{\epsilon_{EV} \cdot \epsilon^{charg} \cdot \epsilon_{elec}^{wtt}}$$

Technical specifications of DV & EV

Saenz et al. [2016], Goeke and Schneider [2015], Murakami [2017]

DVs



©ramtrucks.com

- engine power: 180 kW
- tare weight: 1.5 t
- max. payload: 0.8 t
- fuel cons.: 2-25 l/h

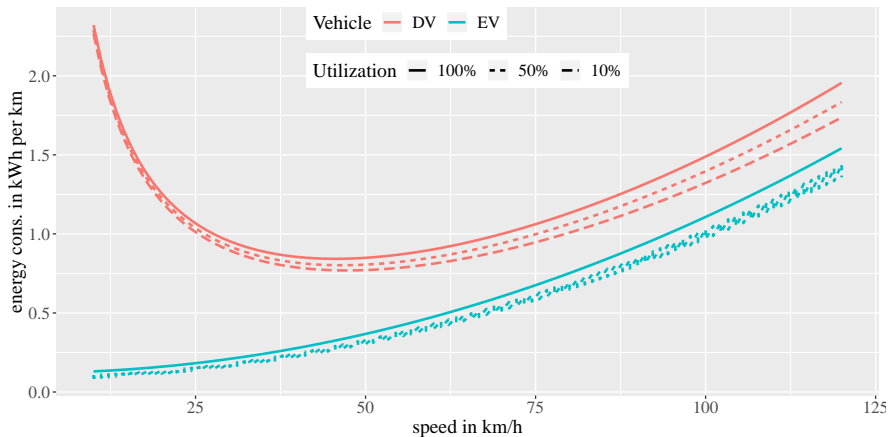
EVs



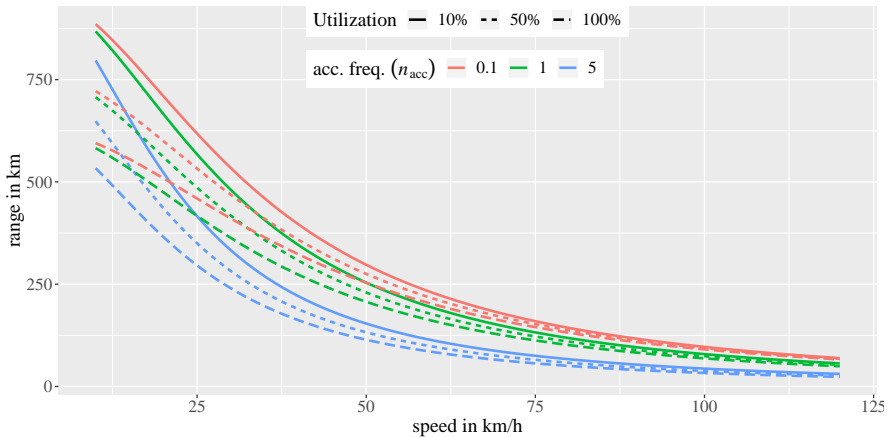
©ups.com

- engine power: 190 kW
- tare weight: 2.0 t
- max. payload: 0.75 t
- battery: 80 kWh

DV & EV energy consumption



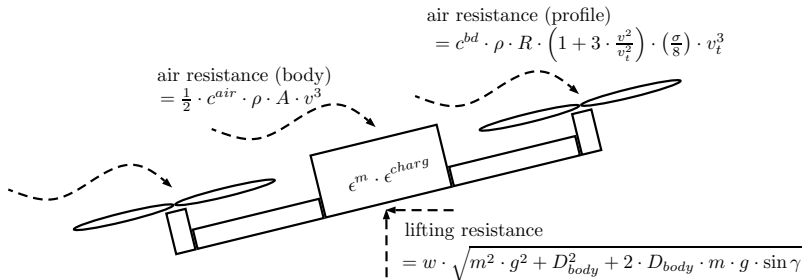
EV range



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Power demand UAVs

Langelaan et al. [2017], D'Andrea [2014], Figliozzi [2017]

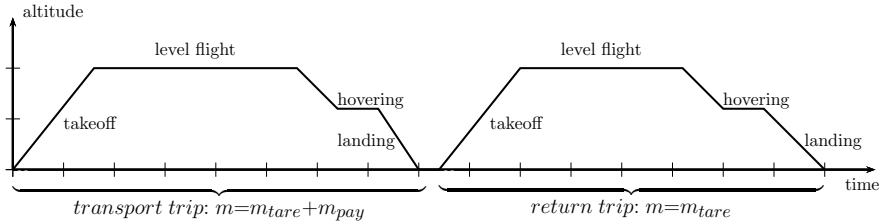


$$P^{UAV}(m, v, \gamma) = P^{air} + \kappa \cdot P^{lift} + P^{profile} + P^{climb} + P^{aux}$$

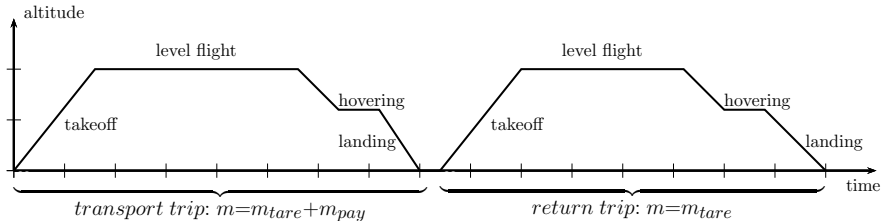
$$E^{UAV}(m, v, \gamma) = \frac{t}{\epsilon^{charg} \cdot \epsilon_{elec}^{wt}} \cdot \left(\frac{P^{air} + \kappa \cdot P^{lift} + P^{profile} + P^{climb}}{\epsilon_{UAV}} + P^{aux} \right)$$

⇒ time depends on speed v , distance d , and wind $v_{head} \rightarrow t = \frac{d}{v - v_{head}} = \frac{d}{v_{net}}$

UAV flight pattern & energy consumption model



UAV flight pattern & energy consumption model

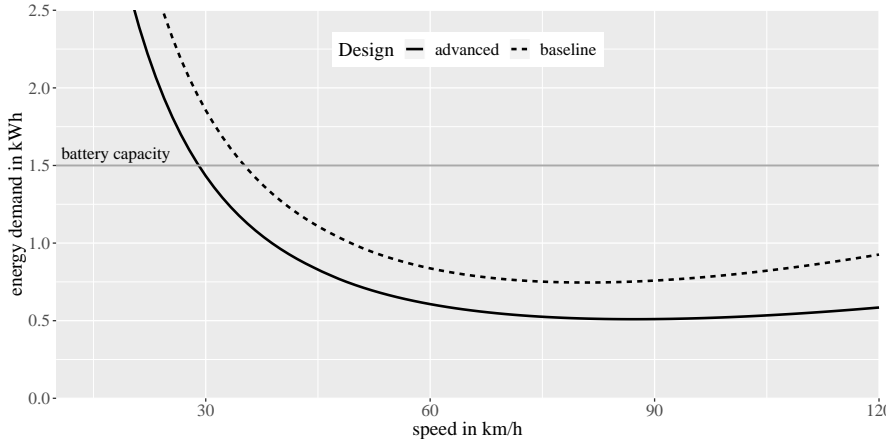


$$E^{UAV} = \frac{1}{\epsilon_{UAV} \cdot \epsilon^{charg} \cdot \epsilon_{elec}^{ttw}} \cdot \left(\begin{array}{l} \text{efficiency correction} \\ t_{tol} \cdot \left(P^{UAV}(m, \nu, 45^\circ) + P^{UAV}(m, \nu, -45^\circ) \right) + \text{take-off and landing} \\ t_{lf} \cdot P^{UAV}(m, \nu, 0^\circ) + \text{level flight} \\ t_{hover} \cdot P^{UAV}(m, |\nu_{head}|, 0^\circ) + \text{hovering} \\ t \cdot \frac{P^{aux}}{\epsilon_{UAV}^{charg}} \text{ auxiliaries} \end{array} \right)$$

$$\implies t = t_{lf} + t_{hover} + 2 \cdot t_{tol} \text{ with } t_{lf} = \frac{d}{v_{net}} - 2 \cdot t_{tol} \text{ and } t_{tol} = \frac{a}{v_{net}}$$

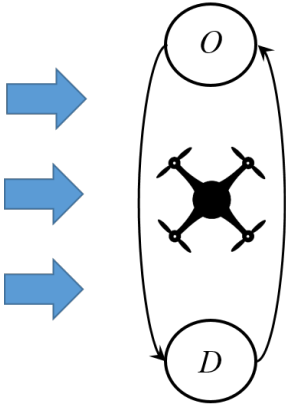
UAV energy consumption depending on speed

idealized trip with 16 km range

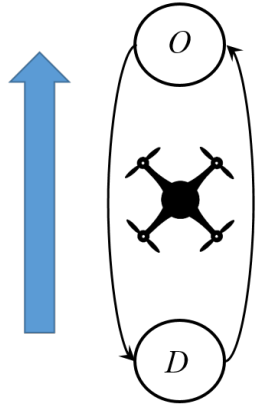


Wind effects

Cross wind

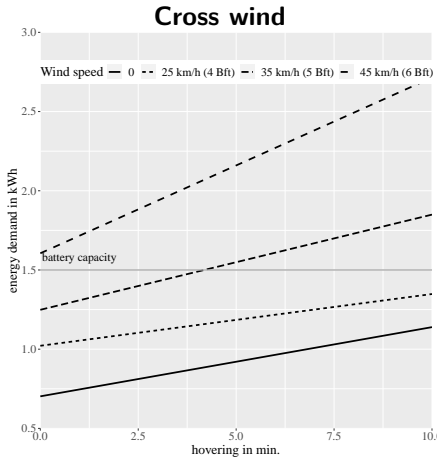


Head/tail wind



UAV energy consumption depending on wind & hovering

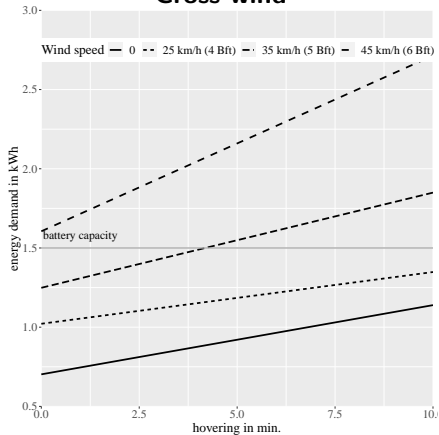
idealized trip with 16 km range



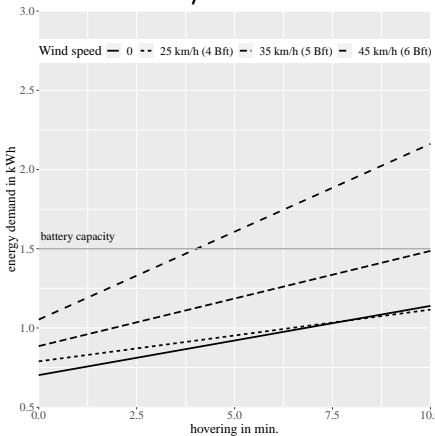
UAV energy consumption depending on wind & hovering

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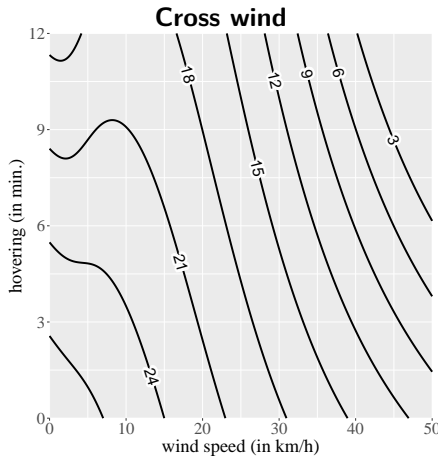
Cross wind



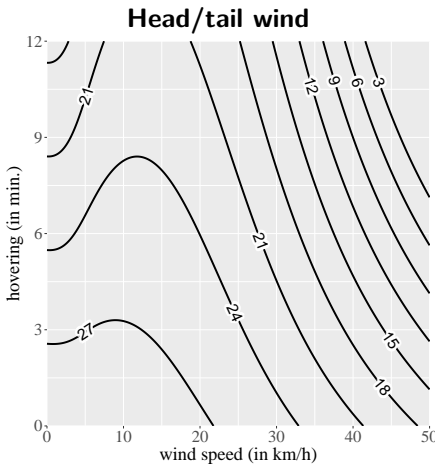
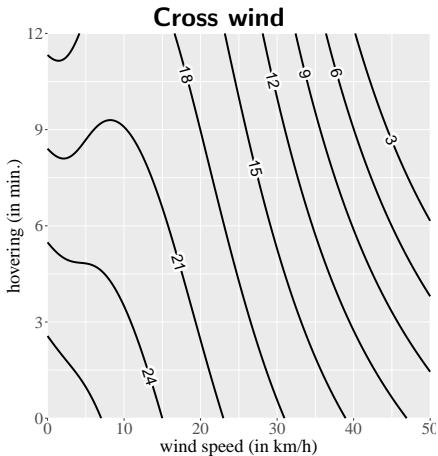
Head/tail wind



UAV range function (in km)



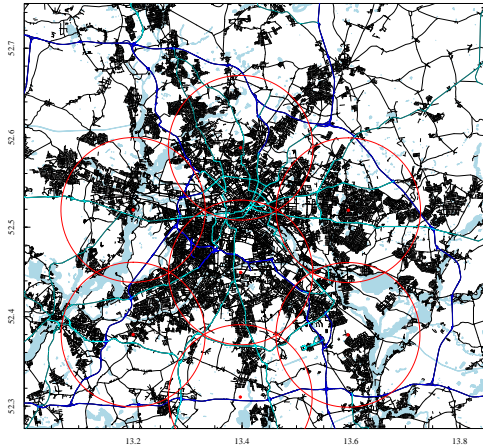
UAV range function (in km)



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Simulation setting - general assumptions

Network data

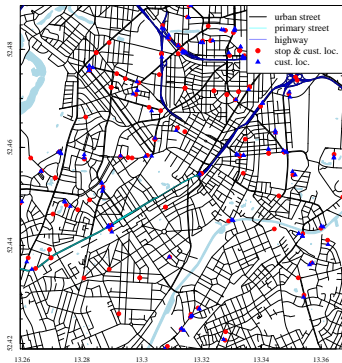


- road network of city of Berlin; 3 road types (highways, primary, residential) differing w.r.t. speed and acceleration frequency
- each customer receives 1 parcel
- each parcel weighs 2.5 kg
- vehicles start from depot
- DVs and EVs use roads; UAVs fly directly
- UAVs hover for 5 minutes
- each UAV can carry 1 parcel
- delivery area with radius 9 km

Simulation setting - experimental design

Customer data

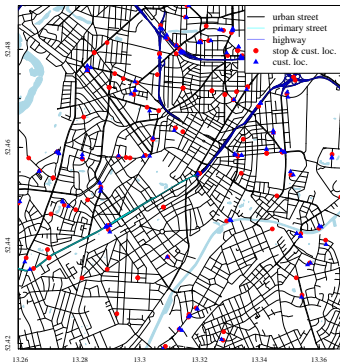
- # customers/tour: [110, 140, 170, 200]
 - # stops/tour: 100
- 1.1 - 2.0 cust./stop
- radius customer area: [2, 4, 6, 8] km



Simulation setting - experimental design

Customer data

- # customers/tour: [110, 140, 170, 200]
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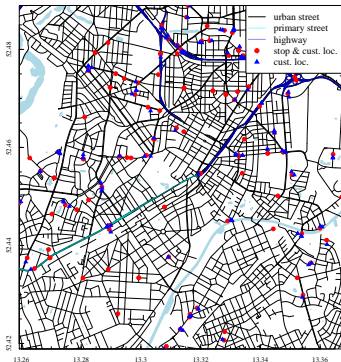
Traffic & wind conditions

level	traffic		wind
	mean speed \bar{v}	acc. freq. n_{acc}	head wind v_{wind}
low	$1 \cdot \hat{v}$	$0.5 \cdot \hat{n}_{acc}$	$N(0, 5)$
medium	$0.95 \cdot \hat{v}$	$1 \cdot \hat{n}_{acc}$	$N(25, 5)$
high	$0.67 \cdot \hat{v}$	$2 \cdot \hat{n}_{acc}$	$N(45, 5)$

Simulation setting - experimental design

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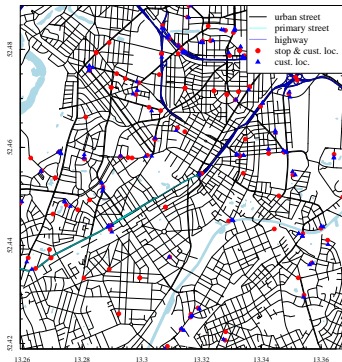
Design:

- full factorial design: 4 factors with 4 levels (cust. data) and 3 levels (env. cond.)

Simulation setting - experimental design

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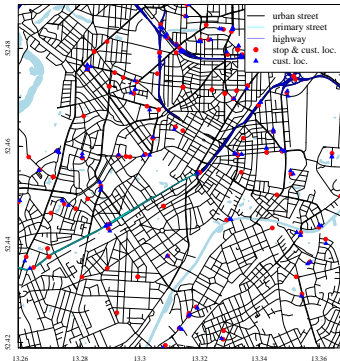
Design:

- full factorial design: 4 factors with 4 levels (cust. data) and 3 levels (env. cond.)
- $4^2 \cdot 3^2 = 144$ settings

Simulation setting - experimental design

Customer data

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Traffic & wind conditions

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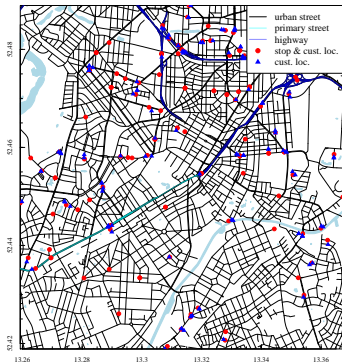
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- $4^2 \cdot 3^2 = 144$ settings
- for each setting: 200 replications

Simulation setting - experimental design

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Traffic & wind conditions

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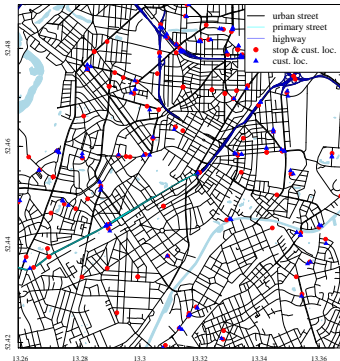
Design:

- full factorial design: 4 factors with 4 levels (cust. data) and 3 levels (env. cond.)
- $4^2 \cdot 3^2 = 144$ settings
- for each setting: 200 replications
- 28,800 instances

Simulation setting - experimental design

Customer data

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Traffic & wind conditions

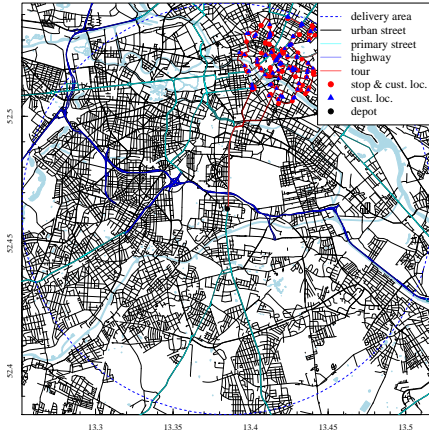
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Design:

- full factorial design: 4 factors with 4 levels (cust. data) and 3 levels (env. cond.)
- $4^2 \cdot 3^2 = 144$ settings
- for each setting: 200 replications
- 28,800 instances
- per instance: solve TSP & calculate WTW energy demands

Delivery area & EV/DV tours (100 stops)

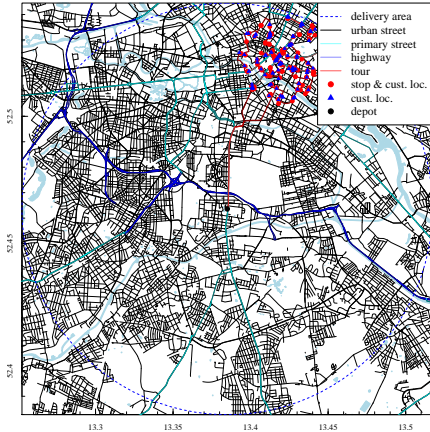
cust. radius: 2 km / # cust.: 200



average tour length: \approx 45-55 km

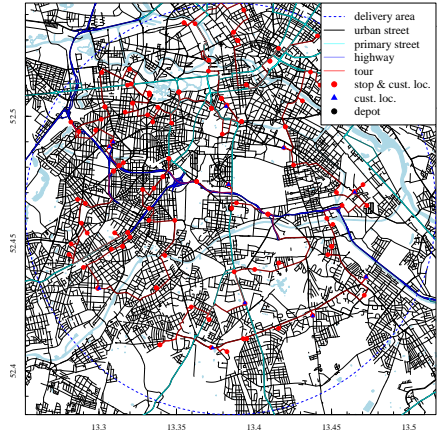
Delivery area & EV/DV tours (100 stops)

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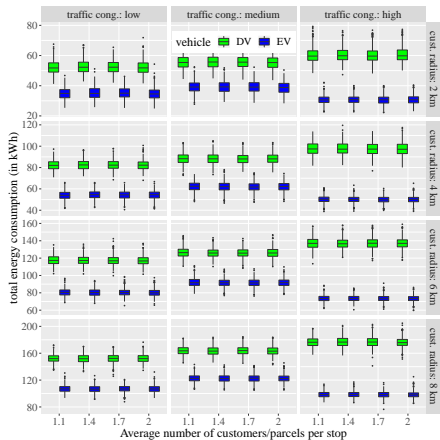
cust. radius: 8 km / # cust.: 110



average tour length: \approx 145-155 km

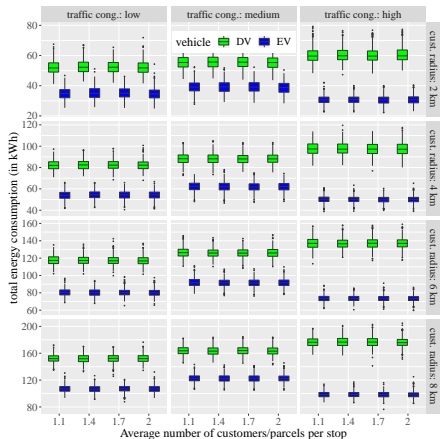
Total energy consumption

Ground-based vehicles only

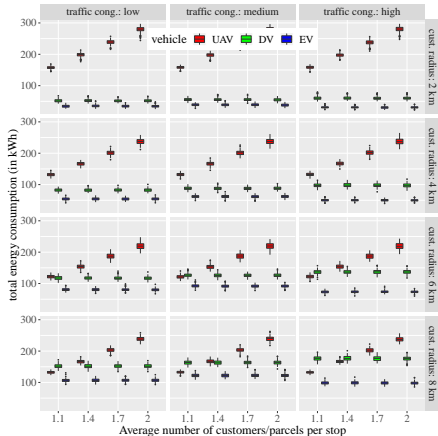


Total energy consumption (medium wind)

Ground-based vehicles only

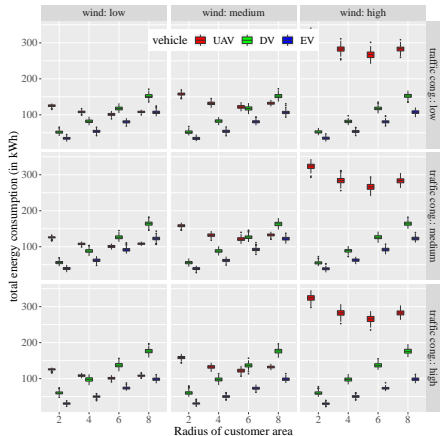


All vehicles



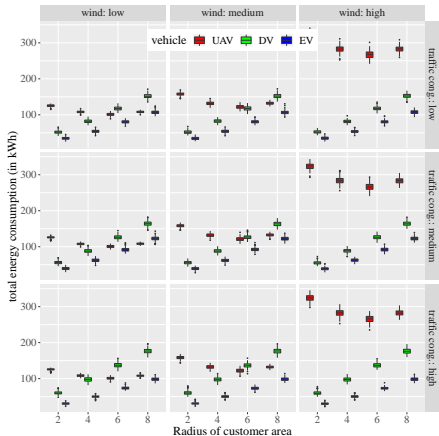
Total energy consumption

1.1 customers/stop

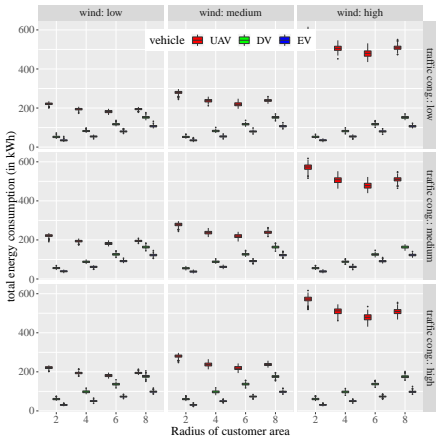


Total energy consumption

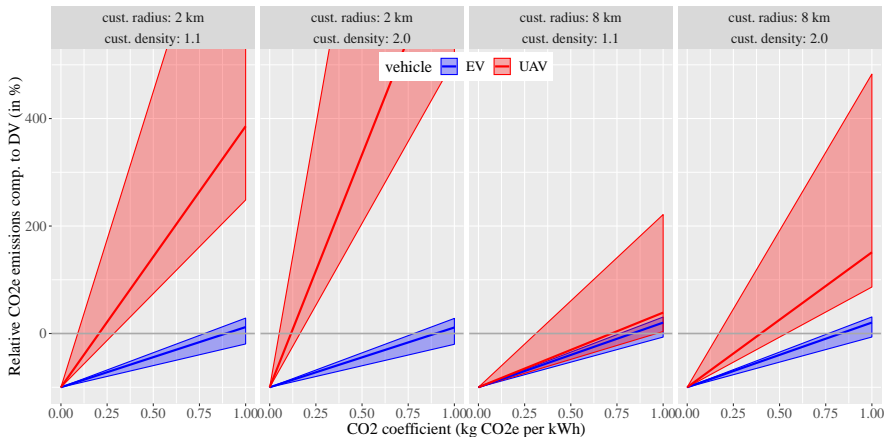
1.1 customers/stop



2.0 customers/stop



GHG reduction potential (compared to DVs)



Summary

- energy demand of drones heavily depends on environmental conditions
 - hovering can be an important aspect for drone delivery systems
 - delivery by drone typically requires more energy than EVs
 - drones require less energy when parcel and customer density
- most probably not useful in cities
- but potentially useful in rural areas
- less energy demand than trucks
 - less dense road infrastructure
 - more predictable weather conditions
 - potentially easier drop-down conditions
 - less regulatory concerns (e.g. due to accident risk etc.)

Finally...

Thanks for your attention.

Kirschstein, T. (2020): Comparison of energy demands of drone-based and ground-based parcel delivery services, *Transportation Research Part D*, 78.

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Simulation parameters

WTT energy efficiency:

Diesel: 90 % ; Electricity: 50 %

Technical vehicle parameters:

parameter	description	unit	DV	EV	UAV		
					curr.	imp.	
A	frontal surface area	m^2		6		0.15	0.15
m_{tare}	tare weight	ton		2.5		0.012	0.012
P	engine power	kW	150	190		—	—
P_{int}	power internal auxiliaries	kW	0.1	0.1		0.1	0.1
f_{idle}	fuel consumption (idle)	l/h	1	—		—	—
f_{full}	fuel consumption (full)	l/h	25	—		—	—
NHV_{diesel}	net heating value	kWh/l	10	—		—	—
cap_{batt}	battery capacity	kWh	—	80		1.5	1.5
z_{batt}	energy density	kg/kWh	—	—		0.15	0.2
ϵ_{eng}	engine efficiency	—	—	—		0.9	0.93
ϵ_{trans}	transmission efficiency	—	—	—		0.9	0.93
ϵ_{char}	charging efficiency	—	—	—		0.9	0.93
n_{rotor}	number rotors	—	—	—		8	8
n_{blades}	number blades	—	—	—		3	3
r	rotor radius	m	—	—		0.4	0.4
c_{air}	air drag	—		0.65		0.5	0.3
c_{roll}	rolling resistance	—		0.008		—	—
c_{bd}	blade drag	—	—	—		0.075	0.075
\bar{c}	rotor mean chord	—	—	—		0.1	0.1
\bar{c}_l	blade lift	—	—	—		0.4	0.4
κ	lifting power markup	—	—	—		1.15	1.15